Joint Winter Runway Friction Measurement Program (JWRFMP)

2000 Testing and Data Analysis

Prepared for
Transportation Development Centre
of
Transport Canada,
National Aeronautics and Space Administration,
and
Federal Aviation Administration

by CDRM, Inc. State College, Pennsylvania, USA and MFT a.s. Oslo, Norway

October 2001



Joint Winter Runway Friction Measurement Program (JWRFMP) 2000 Testing and Data Analysis

by James C. Wambold, PhD J. J. Henry, PhD and Arild Andresen This report reflects the views of the authors and not necessarily those of the Transportation Development Centre of Transport Canada or the sponsoring organizations.

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Un sommaire français se trouve avant la table des matières.

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Transport Transports Canada Canada

PUBLICATION DATA FORM

丁	Canada Canada						
	Transport Canada Publication No.	2. Project No.		3.	Recipient's C	Catalogue No.	
	TP 14062E	9555					
	Title and Subtitle			5.	Publication E	Date	
	Joint Winter Runway Friction Measu 2000 Testing and Data Analysis	rement Program (JW	RFMP):		October	2001	
	2000 Testing and Data Analysis			6.	Performing C	Organization Docume	ent No.
	Author(s)			8.	Transport Ca	anada File No.	
	James C. Wambold, J.J. Henry, and	Arild Andresen			ZCD24	50-B-14	
	Performing Organization Name and Address			10.	PWGSC File	No.	
	CDRM, Inc. P.O. Box 1277	MFT Mobility Friction Oberst Rodes Vei 89			XSD-8-0	01360	
	State College, PA	1165 Oslo	00	11.	PWGSC or T	ransport Canada C	ontract No.
	USA 16804	Norway			T8200-8	3-8593/001/	XSD
	Sponsoring Agency Name and Address			13.	Type of Publ	ication and Period C	Covered
	Transportation Development Centre 800 René Lévesque Blvd. West	(TDC)			Final		
	Suite 600			14.	Project Office	er	
	Montreal, Quebec				A. Bocc	anfuso	
	H3B 1X9				A. Docc	amuso	
	Supplementary Notes (Funding programs, titles of related pure Co-sponsored by NASA, the FAA an		ioti (Dranah of Tr				
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FORMULE DE DONNÉES POUR PUBLICATION

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1.	Nº de la publication de Transports Canada	2. N° de l'étude		 N° de catalogu 	ue du destinataire	
	TP 14062E	9555				
4.	Titre et sous-titre			5. Date de la put	olication	
	Joint Winter Runway Friction Measu	ırement Program (JV	/RFMP):	Octobre	2001	
	2000 Testing and Data Analysis			6. N° de docume	nt de l'organisme e	xécutant
7.	Auteur(s)			8. N° de dossier	- Transports Canad	a
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	James C. Wambold, J.J. Henry et A	ZCD245	U-B-14			
9.	Nom et adresse de l'organisme exécutant	10. N° de dossier	- TPSGC			
	CDRM, Inc.	MFT Mobility Frictio		XSD-8-0	1360	
	P.O. Box 1277	Oberst Rodes Vei 8	96	11. N° de contrat -	TPSGC ou Transn	orts Canada
	USA 16804	College, PA 1165 Oslo				
	USA 10004	Norway		18200-8	-8593/001/X	(SD
12.	Nom et adresse de l'organisme parrain			13. Genre de pub	lication et période v	isée
	Centre de développement des trans 800, boul. René-Lévesque Ouest	Final				
	Bureau 600			14. Agent de proje	et .	
	Montréal (Québec)			Λ Page 6	nofuco	
	H3B 1X9 `		A. Boccanfuso			
15.	Remarques additionnelles (programmes de financement, titr	res de publications connexes, etc.)			
	Projet coparrainé par la NASA, la FA	AA et la Direction de l	a sécurité des a	érodromes de TC	;	
16.	Résumé					
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19.	Classification de sécurité (de cette publication)	20. Classification de sécurité	(de cette page)	21. Déclassification (date)	22. Nombre de pages	23. Prix
	Non classifiée	Non classifiée			xviii, 20, ann	Port et manutention

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ACKNOWLEDGEMENTS

The authors are indebted to Transport Canada for reviewing the report material and giving many valuable comments on the subject material and format of the report, and to Transport Canada and NASA for securing the funding for the Joint Winter Runway Friction Measurement Program (JWRFMP). Also, thanks are due to the many members of the JWRFMP and the aviation community for their assistance and advice.

EXECUTIVE SUMMARY

Measuring the capability of a runway surface to provide aircraft wheel-braking action is fundamental to airport aviation safety, especially under winter conditions. The different seasons, mainly winter, result in the possibility of the runway having contaminants of varying natures and qualities that contribute to reduced braking friction capabilities. A service is warranted for the measurement of winter surface friction, because the operational window for aircraft movement can change quite rapidly and frequently in the winter.

In the past, users of friction information have generally perceived the quality of the friction measurement service as poor. Often, these users have indicated that the reported friction values do not represent the actual braking friction that is experienced with aircraft tire braking.

International research of friction measurement confirmed that friction measurement devices measure and report different friction values for the same surface. Differences occurred among units of the same generic device as well as across different device types. The perception of non-uniformity was compounded by surfaces exhibiting large variances in reported values. These variances further augmented the differences among device types.

Measurements of friction were not calibrated to a common scale in the past. Also, being a non-dimensional ratio of forces, they were never associated with units of a scale, which could be another reason for the resulting differences. Ultimately, dynamic friction measurement results in the highest accuracy, but the procedure is limited to machine component calibrations. Research over the past four years has made significant advances toward solving these problems. Methods of measurement are being improved to increase measurement quality, remove uncertainties, and provide better correlation to aircraft tire braking. Prototype methods that incorporate ground friction measurement devices have shown very promising results.

This study was part of a government/industry project called the Joint Winter Runway Friction Measurement Program, led by the National Aeronautics and Space Administration and Transport Canada. Support is received from National Research Council Canada, the U.S. Federal Aviation Administration, the Norwegian Civil Aviation Authority, and France's Direction générale de l'aviation civile. Organizations and equipment manufacturers from Austria, Canada, France, Germany, Norway, Scotland, Sweden, Switzerland, and the United States are also participating.

Objectives of the project include:

- Compiling a database containing all test data available from ground vehicles and aircraft that participated in the winter and summer runway friction programs.
- Using the data to determine a harmonized runway friction index: the International Runway Friction Index (IRFI).

The objective of this report is to update the 1997-98 JWRFMP report (TP 13836E) with the data collected, analysis and findings through the year 2000.

Statistical IRFI Model

Normally, regression techniques would be used to find relationships between the reported friction values of pairs of ground friction measurement devices. Such a technique assumes that one device's interaction with a surface is similar to another device's interaction with the same surface. The device, or an algebraic transformation of reported friction values, such as the average friction of two or more devices, would be selected as a reference. All devices would then be compared to the reference device to establish transformation constants. A simple linear regression, as shown in the equation below, is seen as a first step or an interim method, which can be applied by the aviation community in the near future. The following equation represents a linear regression of the data for each device to an IRFI reference:

$$\mu_{IRFI} = a + b \times device friction measurement$$

where a is the intercept and b is the gradient that were determined by the regression to the reference device. Past attempts failed because the data were not acquired at the same time in the same wheel track. Also, the sample size was too small. Since 1998, the friction measurement and corresponding data collection have been carried out more systematically. Pairs of measurement devices run in a wave pattern so that they measure the same surface within 15 seconds of each other. However, even with this systematic approach there are considerable variations in the measured surface condition because of the lateral placement of the devices and the resulting effect of surface compaction. The database now includes more than 30,000 friction measurements.

Stability of the Harmonization Method

The correlation constants were calculated for devices that participated in the 1997-98 test seasons and were reported in the 1997-98 JWRFMP report. The constants were calculated by combining the two years of data. However, in the current year, 2000, it was established that not only does a calibration not apply across similar types of devices, but it changes from year to year for a particular device. Figure 1 shows the variations of the IRFI multiplier *b* for the past three years.

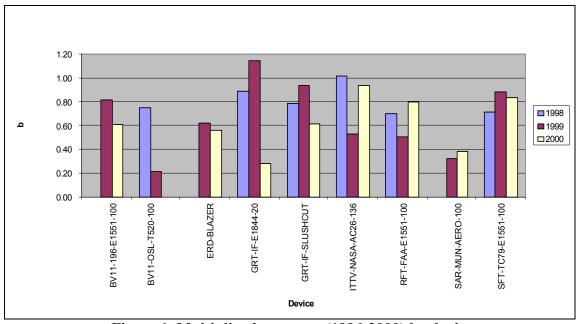


Figure 1. Multiplier b vs. years (1996-2000) by device

Conclusions and Recommendations

The ASTM standard E-2100 defines and prescribes how to calculate IRFI for winter surfaces. IRFI is a harmonized reporting index to provide information to aircraft operators on tire-surface friction characteristics of the movement area.

In addition to reporting surface conditions to aircraft, IRFI can be used by airport maintenance staff to monitor the winter frictional characteristics for surface maintenance actions.

The method evaluates each 100 m (300 ft.) and averages them for each third of the runway. The IRFI method reduces the present variations of the 100 m surface lengths from as much as 0.2 down to typically 0.04. The sampling scheme of a full runway length (spot or continuous measurements) may yield additional variation.

A reference device, which is required for calibration, must be a dedicated device for this purpose only, and the aviation community must agree on its provision, ownership and services. The device chosen for the exercises of the JWRFMP, to demonstrate that IRFI is possible, was an IMAG called the International Reference Vehicle (IRV). The IRV must be evaluated at some point for stability. If it is not stable with time, other references would need to be investigated. All harmonization constants will have to be reworked when a permanent IRFI reference has been designated. In the meantime at least harmonization was demonstrated to work and was accomplished with the devices participating in the JWRFMP.

There is proof that the participating devices in the JWRFMP are not representative of the other devices even when they are of the same generic type. This suggests that

harmonization constants must be determined and applied to individual devices, rather than to generic groups of devices, as was done in the past. To accomplish this, a master device can be calibrated to the IRFI reference device in order to serve as a secondary reference, and the manufacturer or owner of this secondary reference can then calibrate other devices to this master.

Ongoing work has shown that IRFI can be used to predict aircraft braking and will be reported in a separate report. Transport Canada has reported that its version of the IRFI, called the Canadian Friction Index (CRFI), correlates well.

SOMMAIRE

Connaître l'adhérence des pneus d'un avion au freinage est essentiel à la sûreté des opérations aériennes aux aéroports, particulièrement en hiver, alors que les pistes peuvent comporter des contaminants de natures diverses qui réduisent l'adhérence à divers degrés. De plus, en hiver, les conditions de décollage/atterrissage peuvent changer très rapidement et à une fréquence telle qu'il est amplement justifié de constituer un service aéroportuaire de mesurage de la glissance des chaussées.

Par le passé, les utilisateurs de données sur la glissance avaient une piètre opinion du service de mesure de la glissance des pistes. Ces utilisateurs se sont souvent plaints que les valeurs de glissance enregistrées n'avaient rien à voir avec le comportement des pneus au freinage.

Des travaux de recherche sur la glissance des pistes menés à l'échelle internationale ont confirmé que pour une même surface, les appareils de mesure du coefficient de frottement captent et enregistrent des valeurs différentes. Des écarts ont été observés non seulement entre les mesures prises par des appareils de différents types, mais aussi entre les mesures effectuées par un même appareil. Les valeurs enregistrées sur une même surface affichaient de larges écarts, ce qui ne contribuait aucunement à dissiper la perception de non-uniformité et ces fluctuations étaient d'autant plus grandes que différents types d'appareils étaient utilisés.

On ne prenait pas la peine alors de rapporter les mesures du frottement à une échelle commune. De plus, comme ces mesures représentaient un rapport non dimensionnel, elles n'ont jamais été associées aux unités d'une échelle, autre explication possible des écarts enregistrés. Finalement, la mesure du frottement dynamique donne la plus grande précision, mais cette procédure se heurte à une difficulté, soit le calage intégré des éléments de chaque appareil. La recherche menée ces quatre dernières années a grandement contribué à résoudre ces problèmes. Ainsi, grâce au perfectionnement des méthodes de mesure, les résultats sont de meilleure qualité et mieux corrélés avec la performance au freinage des pneus aéronautiques, et les incertitudes sont éliminées. Des méthodes novatrices utilisant des appareils de mesure du frottement au sol ont donné des résultats très encourageants.

Cette étude s'inscrivait dans le cadre d'un projet mené conjointement par le gouvernement et l'industrie, intitulé Programme conjoint de recherche sur la glissance des chaussées aéronautiques l'hiver (PCRGCAH). Transports Canada et la National Aeronautics and Space Administration (NASA) pilotent ce programme, appuyés par le Conseil national de recherches du Canada, la Federal Aviation Administration des États-Unis, l'Administration norvégienne de l'aviation civile et la Direction générale de l'aviation civile de France. Des organismes et des fabricants de matériel d'Allemagne, d'Autriche, du Canada, d'Écosse, des États-Unis, de France, de Norvège, de Suède et de Suisse participent également au programme.

Objectifs du projet :

- Constituer une base de données contenant toutes les données d'essai recueillies par les appareils de mesure au sol et les avions qui ont participé aux campagnes d'essais tenues aussi bien en hiver qu'en été.
- Utiliser les données pour établir un indice harmonisé de glissance des pistes, désigné Indice international de glissance des pistes (IRFI, pour *International Runway Friction Index*).

Ce rapport vise à mettre à jour le rapport de 1997-1998 du PCRGCAH (TP 13836E) à l'aide des données recueillies en 2000, de l'analyse de celles-ci et des résultats subséquents.

Modèles d'IRFI

Habituellement, on se sert de techniques de régression pour établir les relations entre les coefficients de frottement enregistrés par deux appareils différents. Pour utiliser cette technique, on suppose que l'interaction d'un appareil avec une surface s'apparente à l'interaction d'un autre appareil avec la même surface. L'appareil, ou une transformation des coefficients de frottement enregistrés, comme la moyenne des valeurs obtenues par deux ou plusieurs appareils, est choisi comme appareil de référence. Tous les appareils sont alors comparés à l'appareil de référence pour l'établissement des constantes de transformation. Une régression linéaire simple (voir l'équation ci-après) est considérée comme une première étape ou une méthode provisoire que pourraient appliquer les milieux aéronautiques dans un proche avenir. L'équation ci-dessous représente une régression linéaire des données de chaque appareil sur les données d'un appareil de référence IRFI:

 $\mu_{IRFI} = a + b \times coefficient de frottement enregistré par l'appareil$

où *a* est l'ordonnée à l'origine et *b* le gradient, déterminés par la régression sur l'appareil de référence. Les tentatives antérieures de développer un indice uniforme avaient échoué parce que les données étaient recueillies à des moments et à des endroits différents. De plus, les échantillons de données n'étaient pas assez grands. Mais depuis 1998, la mesure des coefficients de frottement et la collecte des données correspondantes sont davantage systématiques. Ainsi, deux appareils de mesure sont lancés l'un à la suite de l'autre, de sorte qu'ils mesurent la même surface à 15 secondes d'intervalle. Mais, malgré cette approche systématique, il subsiste des écarts considérables entre les valeurs obtenues, car le fait de décaler latéralement les appareils produit, au nombre des appareils mis en œuvre, un effet de tassement. La base de données comprend maintenant plus de 30 000 valeurs de mesure du frottement.

Stabilité de la méthode d'harmonisation

On a calculé les constantes de corrélation pour les appareils qui ont servi dans le cadre des essais réalisés en 1997-1998 et dont les résultats ont été présentés dans le rapport 1997-1998 du PCRGCAH. Il a été possible de déterminer les constantes en jumelant les données de deux années. Cependant, en 2000, il a été établi que non seulement les

techniques d'étalonnage diffèrent d'un appareil d'essai à l'autre, mais qu'elles changent également d'une année à l'autre pour un même appareil. La figure 1 présente les écarts du multiplicateur *b* de l'IRFI pour les trois dernières années.

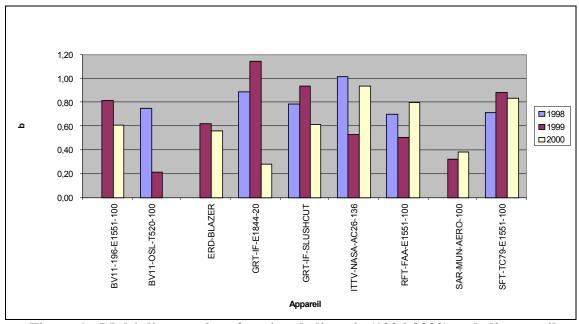


Figure 1 : Multiplicateur b en fonction de l'année (1996-2000) et de l'appareil

Conclusions et Recommandations

La norme E-2100 de l'ASTM prescrit la méthode de calcul de l'Indice international de glissance des pistes (IRFI) en conditions hivernales. L'IRFI est un indice harmonisé destiné à renseigner les exploitants d'aéronefs sur les caractéristiques d'adhérence des pistes.

Cet indice peut également servir accessoirement au personnel d'entretien des chaussées aéronautiques dans le cadre de la surveillance de l'adhérence des pistes en hiver et des activités d'entretien des pistes.

La méthode prescrite par l'ASTM consiste à évaluer la glissance pour chaque 100 m (300 pi) de piste, puis à calculer la moyenne pour chaque tiers de la piste. Cette méthode permet de réduire l'écart obtenu, pour des longueurs de 100 m, d'une valeur qui atteint parfois 0,2 à une valeur se situant généralement aux alentours de 0,04. Le mode d'évaluation de la glissance sur la pleine longueur de la piste (mesures ponctuelles ou continues) peut être une source additionnelle de variation.

Il importe de disposer d'un appareil de référence affecté spécifiquement à l'étalonnage des appareils individuels de mesure. La communauté aéronautique doit déterminer qui en sera le propriétaire, à quels moment s'en servir et quels services il permettra d'offrir. Le PCRGCAH a choisi l'IMAG pour démontrer la validité opérationnelle d'un indice IRFI. L'IMAG est le véhicule de référence actuellement utilisé par le PCRGCAH. Ce véhicule doit d'ailleurs être soumis à une évaluation de sa constance. Si cet appareil n'est pas

constant dans le temps, il faudra étudier la possibilité d'utiliser d'autres instruments de référence. Toutes les constantes d'harmonisation devront être recalculées lorsqu'un indice de référence IRFI sera établi. Entre-temps, on a démontré qu'il était possible d'harmoniser les appareils utilisés dans le cadre du PCRGCAH.

Les essais du PCRGCAH ont montré que les résultats obtenus ne sont pas les mêmes avec tous les appareils de mesure, même s'ils sont du même type. Il y a donc lieu de définir les constantes d'harmonisation et de les appliquer aux appareils individuels, et non au type auquel ils appartiennent, comme cela ce faisait dans le passé. Une des solutions possibles serait d'étalonner l'appareil de référence IRFI sur un appareil primaire qui serait utilisé ensuite comme référence secondaire pour les appareils individuels de mesure.

Les travaux en cours ont établi que l'indice IRFI pouvait être utilisé pour prédire la performance au freinage des aéronefs. Les résultats de ces travaux seront présentés dans un rapport distinct. Transports Canada signale que sa propre version de l'indice IRFI, le Coefficient canadien de frottement sur piste (CRFI, pour *Canadian Runway Friction Index*), est effectivement en corrélation avec l'IRFI.

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DEFINITIONS AND NOMENCLATURE

Acronyms

ASFT Airport Surface Friction Tester - Ystad, Sweden ASTM American Society for Testing and Materials

BV-11 Skiddometer (Bromsvagn "Braking Vehicle"), manufactured by Airport

Equipment Company (AEC), Stockholm, Sweden

CRFI Canadian Runway Friction Index

E-274 E-274 Locked Wheel Tester – K.J. Law and ICC, USA

ERD Electronic Recording Decelerometer
ERD Nissan ERD mounted in a Nissan SUV
ERD Blazer ERD mounted is a Blazer SUV

FAA Federal Aviation Administration, USA

IFI International Friction Index

IMAG Instrument de Mesure Automatique de Glissance, France

IRFI International Runway Friction Index IRV International Reference Vehicle

ITTV Integrated Tire Test Vehicle – NASA, USA

JWRFMP Joint Winter Runway Friction Measurement Program
NASA National Aeronautics and Space Administration, USA
NATAM Norwegian Air Traffic and Airport Management

NRC National Research Council Canada

PTI Pennsylvania Transportation Institute, USA

RFT Runway Friction Tester, manufactured by K.J. Law, Michigan, USA ROAR Road Analyzer and Recorder, manufactured by Norsemeter a.s., Rud,

Norway

RUNAR Runway Analyzer and Recorder, manufactured by Norsemeter a.s., Rud,

Norway

STBA Service Technique des Bases Aériennes, Paris, France

SFT Surface Friction Tester, manufactured by Saab AB, Stockholm, Sweden

SFT-TC79 1979 SFT owned by Transport Canada

Definitions

device configuration, **n.** - a term used to designate the entire test system as used for any friction measurement; it includes, but is not limited to, type of device (force or torque measurements), tire type, size and inflation pressure, slip ratio, normal load and braking system control mode.

base surface, **n.** - the type of surface evaluated. There are four classes: (1) bare pavement dry, (2) bare pavement wet, (3) bare compacted snow, and (4) bare ice.

surface, **n.** - a generic term used in the act of reporting frictional characteristics; it includes the base surface class and the base surface condition.

compacted snow, **n.** - a compressed solid mass of snow that is sufficiently strong to prevent a normally loaded tire operating in a rolling mode from penetrating to the pavement or breaking up the surface.

ice, **n.** - water with or without contaminants frozen into a continuous solid body with or without cracks.

local friction device, **n.** - a particular friction testing device used at a given location to measure friction; the friction values evaluated with this device may be calibrated to IRFI values to provide harmonization.

master friction device, **n.** - a particular friction testing device used at a given location to calibrate local friction devices; the friction values of this device must be calibrated to IRFI values

movement area, **n.** - that part of the airport (aerodrome) used for take-off, landing and taxing of aircraft, consisting of the manoeuvring area and the apron(s).

IRFI reference device, **n.** - a particular friction measuring device selected as a benchmark or reference; it is used to calibrate any local or master friction device to permit local friction device values to be converted to IRFI values for selected base surfaces.

harmonization, **n**. - the transformation of the outputs of different devices used for measurement of a specific phenomenon so that all devices report similar values.

1.0 INTRODUCTION

Measuring the capability of a runway surface to provide aircraft tire braking action is fundamental to airport aviation safety, especially under winter conditions. The different seasons, mainly winter, result in the possibility of the runway having contaminants of varying natures and qualities that contribute to reduced braking friction capabilities. In addition, because the operational window for aircraft movement can change quite rapidly and frequently in the winter, a service is warranted for the measurement of surface friction. The measured results of such services have had serious deficiencies, which have been acknowledged by experts worldwide.

No satisfactory method or technique has been developed to predict the tire braking action of aircraft by using friction data collected by ground vehicles. Only limited, indirect correlations have been achieved between selected ground friction measurement devices and a few aircraft types. One technique that has been used is a grading scale of measured friction values collected by selected panels of pilots and based on past experience of braking action quality. A quantitative relationship between ground friction measurement devices and aircraft tire braking is needed.

Only Canada has a standard, the James Brake Index (JBI), which has been used to predict required runway length. In recent years this index has been revised and renamed the Canadian Runway Friction Index.

1.1 NASA/FAA/TC Joint Winter Runway Friction Measurement Program

The international government/industry initiative, called the Joint Winter Runway Friction Measurement Program (JWRFMP), is being led by the National Aeronautics and Space Administration (NASA) and Transport Canada, with support from the U.S. Federal Aviation Administration (FAA), the Norwegian Civil Aviation Authority, France's Direction générale de l'aviation civile and National Research Council Canada (NRC). Also participating are organizations and equipment manufacturers from Canada, the United States, Austria, France, Germany, Norway, Scotland, Sweden, and Switzerland. The primary objective is to perform instrumented aircraft and ground vehicle tests aimed at improving the safety of aircraft ground operations. One of the program goals is flight crew recognition of less-than-acceptable reported runway friction conditions prior to the "go/no go" or the "land/go around" decision point. With this goal in mind, related studies are being conducted to look at contaminant drag, effects of runway treatments on friction, and, especially, the harmonization of ground vehicle friction measurement. Harmonization will enable friction data to be reported to a unified common index worldwide, which will then be used to predict aircraft braking performance. This report addresses the development of a common harmonized index, called the International Runway Friction Index (IRFI) and its verification through 2000.

A few instrumented test aircraft and a variety of ground friction measurement vehicles were used at several different test sites in North Bay, Ontario, Canada, in 1996 and 1997.

In 1998, testing at Jack Garland Airport, North Bay, Canada, and at Oslo Airport, Gardermoen, Norway, involved special tests and the verification of the IRFI on compacted snow and ice. Testing in 1999 involved the NRC Falcon 20 at North Bay and the NASA 757 at K.Y. Sawyer Airport in Michigan. Ground vehicle testing was conducted at both sites and again at the airport in Gardermoen, and included 11 different ground friction measurement devices (at times with several measurement devices of the same type but with different tires). To date, five aircraft have been used: Dash 8, Falcon 29, and Boeing 727, 737, and 757. Plans for future testing include wide-body aircraft and military cargo aircraft, along with new or improved ground testing equipment

It is expected that dissemination, acceptance, and implementation of the test results throughout the aviation community will be facilitated by several organizations. These include the International Civil Aviation Organization, the American Society for Testing and Materials (ASTM), the Joint Aviation Authority, the International Federation of Air Line Pilots Association, the Air Transport Association, and Airports Council International.

The JWRFMP probably has the most extensive runway friction data ever collected at temperatures of 0°C and below. The data are being added to NASA's tire friction database. Through ASTM Committee E17 on Vehicle-Pavement Systems, work is ongoing to develop a harmonized friction index, or IRFI, which is anticipated to become a standard used by airports to assess the condition of a runway under winter conditions.

The JWRFMP was established to resolve the major elements of the deficiencies stated in Section 1. After three years of testing, with the participation of experts from several countries, a systematic, standardized approach is being developed to achieve harmonized friction measurements. This will lead to a methodology for predicting how aircraft tire braking may behave in response to the most recent reported runway friction properties.

This approach, which is recognized by many as the most viable, was introduced by several speakers at the International Meeting on Aircraft Performance on Contaminated Runways, held in Montreal on October 20-22, 1996. The approach combines some elements of the International Friction Index (IFI), proposed by the World Road Association, with the use of inexpensive digital computing for handling the numerous and detailed pieces of information necessary to reach the objectives of harmonization and better aircraft tire braking predictions.

The results reported in this document will provide comparisons of the different participating testers, with different tires used to measure runway friction for both summer and winter conditions. This is a necessary step to achieve harmonization of different friction measurement devices. In the further development of the IRFI, the International Reference Vehicle (IRV) was introduced in 2000 and is used as the reference to calibrate other testers. The IRV is a special version of the IMAG donated by STBA for that purpose.

Also, it will be necessary to compare the results of IRV to the aircraft tire friction data obtained from the aircraft testing.

2.0 JWRFMP PROGRAM OBJECTIVE

In cooperation with other researchers from Transport Canada, NRC, NASA, and the FAA, the objective is to establish an International Runway Friction Index to harmonize all ground friction measurement so that the common values can be reported and used by airports around the world.

Program Sub-objectives

Compile a database containing all test data available on winter and summer runway friction measurements from different devices and tires, including data on aircraft tire braking performance.

Use the data to develop a harmonized runway friction index.

Report Objective

The objective of this report is to update the 1997-99 JWRFMP reports [2,3] and present the data, analysis and findings through the 2000 test year.

3.0 EQUIPMENT TESTED

A variety of instrumented test aircraft and ground friction measuring vehicles have been used at different test sites in the U.S., Canada, Norway and Germany. The NASA B-737 transport and an NRC Dassault Falcon-20 aircraft were used during January and March 1996 at the Jack Garland Airport in North Bay, Ontario. Seven ground friction measuring devices from six different countries collected comparable friction data for several winter runway conditions including dry, wet, solid ice, dry loose snow and compacted snow.

In the January-March 1997 winter season, similar tests were performed at North Bay with an FAA B-727 transport, the NRC Falcon-20 and a De Havilland Dash-8 aircraft, together with 13 ground friction measuring devices. Data obtained during these investigations helped define the methodology for an IRFI to harmonize the friction measurements obtained with the different ground test vehicles.

In the January-February 1998 winter season, additional data was collected at North Bay, Ontario, with the Falcon-20 and Dash-8 aircraft, together with 11 different ground test vehicles, to further refine the IRFI methodology. Based on the Electronic Recording Decelerometer (ERD), a Canadian Runway Friction Index (CRFI) was established for use by pilots to determine their aircraft stopping distance under compacted snow and ice conditions. In March 1998 several different ground friction measuring devices

participated in conducting nearly 800 test runs under compacted snow- and ice-covered surface conditions at a new test track facility located at Gardermoen Airport near Oslo, Norway.

During the January-March 1999 winter season the Falcon-20 aircraft and ground vehicle data was collected at North Bay. Also, in 1999 a NASA B-757 aircraft and ground vehicle data was collected at a new test site, Sawyer Airbase in Gwinn, Michigan. These tests were followed with additional ground vehicles (9 different devices) that obtained friction data at the Ottar K. Kollerud test track at Gardermoen Airport in Norway. Data from these tests was used to further refine and improve the IRFI methodology and define the present correlation constants in the IRFI standard. It is interesting to note that under similar runway conditions at these three different test sites, friction data from the same ground vehicles tested at all three sites were in close agreement and the IRFI methodology was further substantiated.

During the January - March 2000 winter season, one week of testing at North Bay, Ontario, involved the Falcon-20 aircraft and 10 ground friction measuring vehicles. Tests with an Aero Lloyd A320, a Sabena Airlines A320, a Deutsche British Airways B-737-300 and a Fairchild/Dornier 328 aircraft were conducted at Munich Airport, Germany, February 21-25, 2000. Thirteen ground test vehicles participated in the Munich testing. In 2000, 60 test runs were conducted with five aircraft and over 1000 runs were completed with the ground vehicles.

Three years of NASA Aircraft Tire/Runway Friction Workshop data (1998-2000) have been combined with data from fifteen weeks of winter testing at North Bay, Ontario (1996-2000), one week at Sawyer Airbase, Gwinn, Michigan (1999), two weeks at Oslo, Norway (1998-99), and one week at Munich, Germany (2000).

In summary, the number of runs and segments runs made year by year, since 1998, are given in Figure 1 as a bar chart. Segments are typically 100 m sections and in most cases there were three segments in a run. Figure 2 gives the number of runs and segment runs made by site.

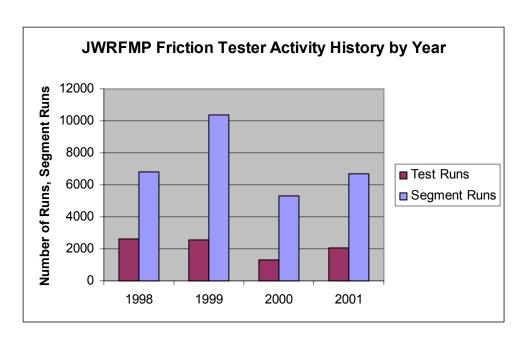


Figure 1. Number of runs and segment runs made by year

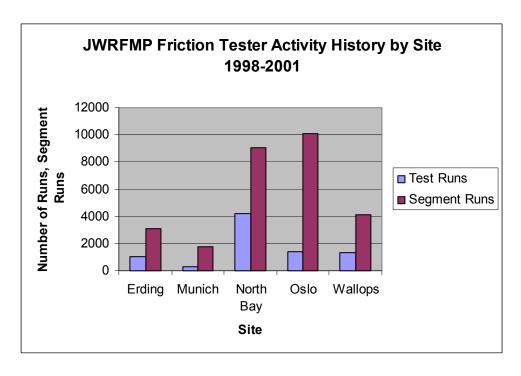


Figure 2. Number of runs and segment runs made by site

Since the beginning of the JWRFMP in January 1996, nine aircraft and 40 different ground devices have collected friction data at North Bay, Ontario; Sawyer Airbase, Gwinn, Michigan; NASA Wallops Flight Facility, Virginia; Oslo, Norway; and Munich, Germany. A total of 442 aircraft runs and over 11,000 ground vehicle runs were

conducted on nearly 40 different runway conditions. More than 300 individuals from nearly 50 organizations in 12 countries have participated with personnel, equipment, facilities and data reduction/analysis techniques. The CRFI and the IRFI are two major outcomes from these efforts to harmonize ground vehicle friction measurements and to identify the relationship to aircraft stopping performance. Two international aviation conferences have been held in Montreal (Oct. 1996 and Nov. 1999) to disseminate the test results and obtain recommendations for future testing. Data from the seven annual NASA Tire/Runway Friction Workshops have been successfully completed to add dry and wet surface ground vehicle friction data to the database. Efforts were initiated in 2000 not only to get funding support from the European Union, but also to get expanded support from the aircraft manufacturers and the airlines. Dialogue to obtain assistance from the International Civil Aviation Organization, the Air Line Pilots Association, and Airports Council International will continue.

A substantial friction database has been established, with both ground vehicle and aircraft winter friction measurements. For each friction value, the database provides the name/type of device, test location, speed, tire specifications, surface conditions and ambient weather conditions. Table 1 is a list of all of the aircraft that have run tests in the JWRFMP and Table 2 is a list of all of the ground friction devices that have participated in the JWRFMP.

Table 1. List of test aircraft that participated in the JWRFMP, 1996 to 2000

AIRCRAFT TYPE	OWNER/OPERATOR	MANUFACTURER
Falcon-20	National Research Council Canada	Dassault Aircraft Company
B-737-100	NASA Langley Research Center	Boeing Commercial Airplane Group
B-727-100	FAA Technical Center	Boeing Commercial Airplane Group
Dash-8	DeHavilland Aircraft Company	DeHavilland Aircraft Company
Dash-8	NAV CANADA	DeHavilland Aircraft Company
B757-200	NASA Langley Research Center	Boeing Commercial Airplane Group
A320	Aero Lloyd	Airbus Industrie
A320	Sabena Airline	Airbus Industrie
B-737-300	Deutsche British Airways	Boeing Commercial Airplane Group

At all test sites, NRC provided an ice and snow specialist who classified the winter contaminate. Typically he measured the water content, density, air and surface temperature, and depth of the contaminate. He also provided observations on the tire tracks produced by the test aircraft and ground vehicles. This data along with the hourly flight weather has also been included in the database

Table 2. Ground friction devices that participated in the JWRFMP, 1996 to 2000

Owner	Device Name	Notes	Manufacturer
Airport Surface Friction Tester AB, Sweden	Airport Surface Friction Tester Ford Taurus		Airport Surface Friction Tester AB, Sweden
Airport Surface Friction Tester AB, Sweden	Airport Surface Friction Tester Generic		Airport Surface Friction Tester AB, Sweden
Oslo Airport, Norway	Airport Surface Friction Tester SAAB 9-5		Airport Surface Friction Tester AB, Sweden
Airport Surface Friction Tester AB, Sweden	Airport Surface Friction Tester SAAB 9-5C		Airport Surface Friction Tester AB, Sweden
NASA Langley Research Center	BOWMONK mounted in Blazer		Bowmonk, United Kingdom
FAA Technical Center	BV-11 Trailer		Airport Equipment Company, Sweden
Oslo Airport, Norway	BV-11 Trailer		Airport Equipment Company, Sweden
Vienna Airport, Austria	BV-11 Trailer Vienna Airport		Airport Equipment Company, Sweden
Zürich Airport, Switzerland	BV-11 Trailer Zurich Airport		Airport Equipment Company, Sweden
NASA Langley Research Center	Diagonal Braking Vehicle		NASA Langley Research Center, USA
Transport Canada	ERD mounted in Chevrolet Blazer		Transport Canada, Canada
Transport Canada	ERD mounted in NISSAN Van		Transport Canada, Canada
Transport Canada	ERD mounted in truck Staff23 North Bay		Transport Canada, Canada
Transport Canada	ERD-179 mounted in Chevrolet Blazer		Transport Canada, Canada
Transport Canada	ERD-234 mounted in Chevrolet Blazer		Transport Canada, Canada
Irvine Findlay Inc., Scotland	Griptester Trailer		Irvine Findlay Inc., Scotland
Department of National Defence, Canada	Griptester Trailer		Irvine Findlay Inc., Scotland
Norwegian Air Traffic and Airport Management	Griptester Trailer		Irvine Findlay Inc., Scotland
French Civil Aviation Administration	IMAG Trailer		S.T.B.A Airports, France
NASA Langley Research Center	Instrumented Tire Test Vehicle		NASA Langley Research Center, USA
French Civil Aviation Administration	IRFI Reference Vehicle Trailer		S.T.B.A Airports, France
Ministry of Transportation, Ontario	Norsemeter ROAR Trailer		Norsemeter AS, Norway
Department of Transportation, Iowa	Norsemeter SALTAR		Norsemeter AS, Norway
Norwegian Road Research Laboratory, Oslo	Optimum Surface Characteristics Analyzer Recorder		Norsemeter AS, Norway
Norwegian Air Traffic and Airport Management	RUNAR Prototype Trailer		Norsemeter AS, Norway
FAA Technical Center	Runway Friction Tester		K.J.Law Engineers, Inc., USA
Munich Airport, Germany	SARSYS SAAB 9000 Mrk V3		SARSYS, Sweden
Dusseldorf Airport, Germany	SARSYS SAAB 9-5C, Ser # 813		SARSYS, Sweden
FAA Technical Center	Surface Friction Tester		SAAB GM, Sweden
Transport Canada	Surface Friction Tester SAAB 1979		SAAB GM, Sweden
Transport Canada	Surface Friction Tester SAAB 1985		SAAB GM, Sweden
Transport Canada	Surface Friction Tester SAAB 1985 Turbo		SAAB GM, Sweden
Hannover Airport, Germany	Surface Friction Tester		SARSYS, Sweden
NASA Langley Research Center	Tapley meter mounted in Blazer		Tapley, Canada
Pennsylvania State University, PTI	ASTM E-274 2 wheel Trailer	Wallops Only	Pennsylvania State University, USA
Pennsylvania State University, PTI	ASTM E-274 Trailer Mk III	Wallops Only	Pennsylvania State University, USA
Department of Transportation, Virginia	ASTM E-274 Trailer	Wallops Only	International Cybernetics, USA
Department of Transportation, Virginia	British Pendulum Tester	Wallops Only	W.F. Stanley, United Kingdom
Federal Highway Administration	British Pendulum Tester	Wallops Only	W.F. Stanley, United Kingdom
Pennsylvania State University, PTI	British Pendulum Tester	Wallops Only	W.F. Stanley, United Kingdom
Nippo Sangyo Co.,Ltd	Dynamic Friction Tester	Wallops Only	Nippo Sangyo Co., Ltd., Japan
Generic device	Mu-Meter Trailer	Wallops Only	Douglas Equipment Company, United Kingdom

4.0 DEVELOPMENT OPTIONS CONSIDERED

Two approaches were considered as models, a statistical model and a physical model valid for defined surface classifications. In both approaches, runway measurements are associated with surface segments of the runway. The statistical method differentiates between segments that are winter-contaminated versus wet or dry, whereas the physical method separates segments that have the same surface classification, and harmonization is performed on a segment by segment basis, applying the appropriate harmonization constants for the surface class. Because further development of the physical model would require texture data and analysis from full-length, operational runways and since the statistical model was working satisfactorily, further work on the physical model was suspended in 2000. It would be a worthwhile effort in the future to continue studying the data collected to establish the physical parameters so that reported friction values can be normalized before applying the statistical harmonization method. This should lead to the reduction of errors.

The correlation constants for the statistical model were calculated for devices that participated in the 1998-1999 test seasons and were reported in the 1997-99 JWRFMP reports [2, 3]. The constants were calculated by combining the two years of data. However, in the current year, 2000, it was established that not only does a calibration not apply across similar types of devices, but it changes from year to year for a particular device. Thus, correlation constants are now calculated on a year by year base. Section 6 provides a more detailed discussion on device stability and reproducibility.

5.0 ESSENTIAL ELEMENTS OF THE STATISTICAL HARMONIZATION METHOD FOR IRFI

Normally, regression techniques would be used to find relationships between the reported friction values for pairs of devices. One device, or an algebraic transformation of reported friction values, such as the average friction of two or more devices, would be selected as a reference. All devices would then be compared with the reference device to establish transformation constants. The model assumed that when the interaction of one measurement device with one surface changed, all other similar tire-surface interactions would change in a similar way under the same conditions.

The statistical model provides good correlations with reasonable standard errors for bare ice and bare compacted snow surfaces, with the advantage that it is not necessary to identify the exact class of snow or ice contaminating the surface. For bare dry pavement and bare wet pavement, another set of correlations must be used. In addition, texture information or speed gradient is needed in the correlation equation for bare dry and bare wet pavement. For bare wet pavement, the International Friction Index as specified in ASTM Standard E-1960 is recommended.

The field test data sampling for the model includes both ice and snow surfaces in order to create a data set of sufficient range to enable linear regressions.

A simple linear regression, called the statistical IRFI, is seen as a first step or an interim method that can be applied by the aviation community now. This model is a linear regression of the data for each device to a (virtual) IRFI reference:

$$IRFI = a + b \text{ x device friction measurement,} \tag{1}$$

where a is the intercept and b is the gradient, and where these constants were determined by regression with the reference device. Past attempts failed because the data used were not collected at the same time in the same wheel track. In 1998, the data were collected more systematically: pairs of measurement devices made each run consecutively, in a wave, so that they measured the same surface within about 15 seconds of each other. Previous data were not collected in this manner, and it was found that the surface characteristics could change so quickly that the different measurement devices had actually tested different surfaces and so the regression analysis was not valid. This change in time is critical when regressions are being made, but not as critical for operating conditions.

5.1 IRFI Reference Selection

A true value is needed in order to perform a linear regression; therefore, a virtual device called the reference was developed from combinations of devices for the 1998-1999 years. Based on the review [2,3] it was concluded that the best option for the reference was to use the average of the SFT-TC79 and the IMAG. However the SFT-TC79's instrumentation was updated in 1999, making it appear as another device, and the virtual device reference was dropped. In late 1999, STBA offered a second and dedicated IMAG to the JWRFMP and it was accepted and designated as the International Reference Vehicle (IRV) for the JWRFMP. The IRV is now dedicated to the project and not used for any other purpose. A separate study was performed to relate the IMAG used in 1998, 1999 and 2000 to the IRV [4]. This study concludes that the IRV = 0.95 * IMAG. Thus the reference now used for calibration is IRV or 0.95*IMAG, if IRV data is not available.

5.2 IRFI Correlations

The three tables in Appendix A give the IRFI correlation constants *a* and *b* for each of the years 1998, 1999 and 2000. In 1998 the *a* values ranged from -0.05 to 0.08 with an average of 0.03. In 1999 the *a* values ranged from 0.00 to 0.17 with an average of 0.09, and in 2000 the *a* values ranged from 0.04 to 0.25 with an average of 0.15. Similarly the *b* value varied from 0.70 to 1.01 in 1998 with an average of 0.82, from 0.21 to 1.14 in 1999 with an average of 0.67, and from 0.28 to 0.99 in 2000 with an average of 0.62.

5.3 Errors of Fitted IRFI Values

Also given in the three tables in Appendix A are the correlation R² and the standard error of estimate for each of the years 1998, 1999 and 2000. In 1998 the R² ranged from 0.45 to 0.99 with an average of 0.86. In 1999 the R² ranged from 0.05 to 0.74 with an average of 0.46, and in 2000 the R² ranged from 0.10 to 0.99 with an average of 0.62. Similarly, the standard error of estimate varied from 0.023 to 0.076 in 1998 with an average of 0.047, from 0.031 to 0.062 in 1999 with an average of 0.045, and from 0.023 to 0.096 in 2000 with an average of 0.059.

In looking at these values, it appears the correlations were not as good in 1999 and 2000 as in 1998. On the average this is true for several reasons. In 1998 extra care was exercised in a number of the field tests to ensure no loose snow was present on the bare compacted snow and bare ice surfaces. In 1999 the tests included tests in deep snow and more tests were conducted with some loose snow on the ice and packed snow, making the sites more variable and thus subject to lateral position of each device. In 2000, tests were conducted when the conditions were very poor due to lack of snow and the test beds were very variable. This shows the need for good test conditions to maintain the best accuracy.

It should also be noted that devices tested at all sites generally had better R^2 and standard error of estimate than those just tested in Europe. Even so, the average standard error of estimate was less than 0.05 and more than half of the devices were lower. This is in comparison to as much as 0.2 without the IRFI harmonization applied.

5.4 Errors of Predicting IRFI Values

Due to the natural scatter in friction values typically obtained on a runway surface, the predicted IRFI value will show a similar scatter when harmonization is applied to individual reported friction values by a local airport device. The harmonization method is not designed to moderate any surface variability or take into account local runway variability.

The pairs of data samples collected to determine a harmonization equation has variability about the fitted equation line, often expressed in standard deviation. The prediction interval for a given confidence level is proportional to this standard deviation. In other words, the range in error when calculating IRFI values for a harmonized device is a characteristic of the original paired data collection for the determination of the harmonization equation.

It is therefore not possible to calculate what errors the IRFI values would have at a local airport runway that was not part of the original paired data collection.

One may, however, venture to state that provided that the harmonization paired data collection has a sufficient range in friction levels and surface textures and includes representative operational runway characteristics, the error would be within the bounds of the harmonization data set variability. This variability is largely surface variability. Such bounds have been found typically in JWRFMP data sets to be in the order of +/- 0.10 friction units for a 95% confidence level, i.e. 19 of 20 calculations will be within an error of 0.10 friction value. Most of this error is due to surface variability. One may therefore argue that these bounds are not relevant for the friction values of harmonization transforms, since they largely stem from surface variability. The fitted harmonization transform is a product of averaging out much of the surface variability to find the quantitative relationship between two devices.

5.5 Limitations of and How to Improve the Statistical IRFI

No correlations can be expected to remain stable with time since, for example, the devices change, new tires are installed, and the equipment is subjected to wear. Thus, there is a need to have periodic correlations to maintain accuracy.

The exercise performed with a chosen reference demonstrates that harmonization can be achieved with a statistical model. The issue of making available a permanent reference device for the airport industry was solved with the donation of the IRV by STBA. However, there is still a need to evaluate the reference device to aircraft. Based on this evaluation, there may still be a need to design and build a special reference device. With this in mind, ASTM Committee E17 has formed a working group to address this possibility.

The IRFI has initially been studied as a common unit of friction measure. When bringing the IRFI transforms into practical use at airports that have different sampling techniques of their runways, it must be expected that the practical implementations will diverge in reported IRFI values. Notably, continuous friction measuring devices sample contaminated and non-contaminated sections of a runway and include these sections in the harmonization. A spot measuring device may collect only selected contaminated sections of the runway. The IRFI was not designed to overcome differences in sampling techniques.

6.0 STABILITY OF THE HARMONIZATION METHOD

6.1 Reproducibility Concerns

When several friction measuring devices of the same standard type are brought together to measure the same surface object, the degree with which they report the same value of friction is called reproducibility. Any differences in reported friction values across the

devices can be expressed in terms of standard deviation or standard error relative to the arithmetic means of all the measures from all devices studied.

Recent and unique studies performed by the Norwegian Air Traffic and Airport Management as described in [5 to 8] have demonstrated that reproducibility of two different kinds of continuous friction measuring devices was 0.05 friction units for both kinds operated at 65 km/h. This was achieved when the devices were in a technical state as normally used at Norwegian airports. Every effort was made to operate the devices under equal conditions during the field testing. The studies included 25 and 15 units, respectively, of standard GripTesters and non-standard BV11s configured with ASTM smooth measuring tires. The measurements were made under self-wet conditions on a total of 32 surface segments of 100 m each, made of 8 different asphalt mixtures. The macrotexture of these recipes ranged from 24 km/h to more than 260 km/h in IFI speed numbers, corresponding to 0.3 to 2.5 mm mean texture depth as measured by the sand patch method according to ASTM E-965. The friction values were averages of three runs across each segment by each device.

After thorough machine part inspections, replacements of out-of-tolerance worn parts, instrumentation calibration by the manufacturer, and fitting of new measuring tires, the reproducibility of the GripTesters was improved from 0.05 to 0.03 friction units in terms of standard deviation as shown in Figure 3. A similar exercise was not performed for the BV11s as shown in Figure 4.

It is believed that a significant part of the 0.03 value of reproducibility stems from surface and field test variability. The devices were not measuring exactly the same tracks and had different host vehicles and drivers. The self-wet systems had no feedback control of the water flow. However, the figure should be taken as an indication of what the reproducibility in terms of standard deviation can be at its best for a cross section of asphalt surfaces. It may be more prudent in many evaluations to use the 0.05 figure, as first presented above, as representative of operational equipment states, when equipment is partly worn and fitted with partly worn measuring tires.

Reproducibility: Average Friction Value by Device at 65 km/h on 8 Ashphalt Types

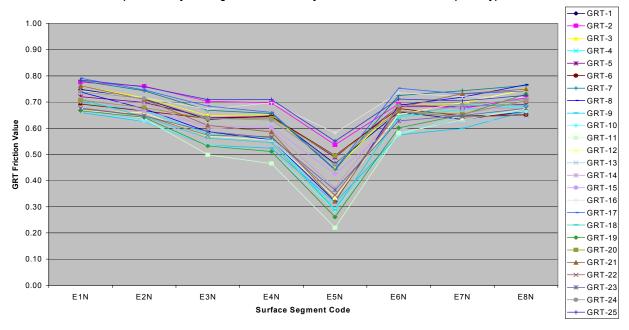


Figure 3. A sample extract of the NATAM database [5, 6, 7, 8] for GripTester devices. Each friction value shown by marker is the average of three runs.

Reproducibility: Average Friction Values by Device at 65 km/h on 8 Asphalt Types

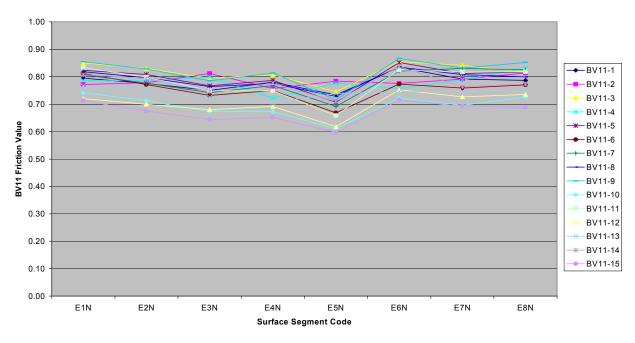


Figure 4. A sample extract of the NATAM database [5, 6, 7, 8] for BV11 devices. Each friction value shown by marker is the average of three runs.

6.2 Time Stability of Individual Devices

In order to evaluate the time stability of the individual devices, a year-by-year comparison of the IRFI constants in JWRFMP was made. Appendix A gives the values of the IRFI constants a and b for each of the years 1998, 1999 and 2000. In addition the regression \mathbb{R}^2 , standard error, number of data points and some comments are given for each device. The year-by-year regressions also show that the same types of devices can produce very different results that require different IRFI regression constants. The tables clearly show that not only are there differences within a class of devices, but that an individual device changes from year to year. To show this, a bar chart of how the multiplying constant b varies is given in Figure 5. Only devices that were calibrated for two or more years in a row are included.

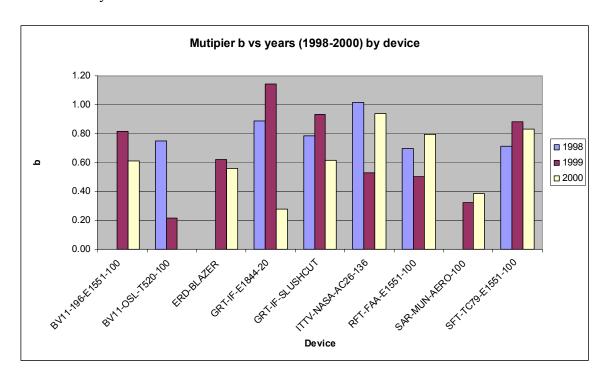


Figure 5. IRFI multiplier constant b versus years 1998 to 2000

Based on the findings from the research of 2000, the ASTM standard was modified to require annual determination of the IRFI harmonization coefficients.

7.0 THE ASTM E 2100 STANDARD

This section describes the method developed and standardized by ASTM in the year 2000. The standard's number and title are: ASTM E 2100-00, Standard Practice for Calculating the International Runway Friction Index. A separate Transport Canada report [9] provides more detail, a summary of which is given here.

The local friction device can be harmonized in one of two ways: by conducting field testing with the IRFI reference device or with a secondary harmonized device called a master device. The method of using a secondary harmonized device was introduced by the standards committee because of practical reasons such as a limited availability of the IRV to all regions of the world and the costs of bringing local devices to field test sites with the only IRV. A method of secondary harmonized references had not been researched by the JWRFMP prior to this.

The field test collects friction data for each surface class for which the local device can be used. When a local friction device has different selectable modes of operation (for example, fixed or variable slip measurement), each mode of operation is treated separately. The local friction device is operated according to the manufacturer's instructions for the device and run within the range of speeds for which it is to be harmonized. If there is a standard test method for the device, it should also be followed.

7.1 Method 1-Harmonization with the IRFI Reference Vehicle

The local device is harmonized to report an IRFI by measuring friction on surfaces with the IRFI Reference Vehicle. A minimum of 8 surfaces covering a range of friction values from 0.1 to 0.7 as measured by the IRFI Reference Vehicle shall be included. Harmonization constants a and b are determined for the speed at which the local device normally operates. Test speeds shall be maintained within ± 3 km/h (1.6 knots, 2 mph). The measurements with the local friction device and the IRFI reference device shall be taken on a segment within 2 minutes of each other.

Linear regressions are as follows:

$$FR_{ref} = a + b \cdot FR_{local}, \tag{2}$$

where FR_{ref} is the friction value reported by the reference device and FR_{local} is the local device measured value. The harmonization constants for the device are a and b. The correlation coefficient of the regression and the standard error of estimate shall be reported. Typical values for devices that have been harmonized are given in the tables of Appendix A. These results were for specific local devices that were harmonized in the JWRFMP. They are not applicable to other local friction devices or to other test speeds, which must be calibrated with the device configuration for that device.

Subsequent measurements made by the local friction device can be harmonized using the regression constants of the device:

$$IRFI = a + b \cdot FR_{local} \tag{3}$$

Whenever the local friction device is modified, repaired or recalibrated, new harmonization constants shall be determined. Note: Many operator handbooks use the term calibration for set-up, including adjustments to dynamometers or weight scales of the machine prior to measurements. In this report recalibration is associated with the

replacement of sensors (strain gauges) or other mechanical-electronic parts of the instrumentation that have a multi-year service life.

7.2 Method 2-Harmonization with a Master Device

The local device is harmonized to report an IRFI by measuring friction on surfaces with a master device that has been calibrated to the IRFI reference device. A minimum of 8 surfaces covering a range of friction values from 0.1 to 0.7 as measured by the master device shall be included. Harmonization constants (a'', b'') shall be determined for the speed at which the device normally operates.

The master device is harmonized by measuring friction on several base surfaces with the IRFI reference device. All surfaces shall be included. A minimum of 5 repeated runs on 8 surfaces covering a friction range from 0.1 to 0.7 as measured by the IRFI reference device are to be included. The harmonization constants (a', b') are determined at speeds at which the device normally operates. Test speeds shall be maintained within ± 3 km/h (1.6 knots, 2 mph).

The measurements with the local friction device and the master device, and for the master device with the IRFI reference device shall be taken on a segment within 2 minutes of each other.

A linear regression is of the form:

$$FR_{ref} = a' + b' \cdot FR_{master} \tag{4}$$

$$FR_{\text{master}} = a'' + b'' \cdot FR_{\text{local}}$$
 (5)

Substituting equation 5 into equation 4 gives:

$$FR_{ref} = a' + b' \bullet (a'' + b'' \bullet FR_{local}), \tag{6}$$

Then:
$$a = a' + b' \cdot a''$$
 and $b = b' \cdot b''$, (7, 8)

where FR_{ref} is the friction value reported by the reference device for each 100 m segment, FR_{master} is the master device measured value for each 100 m segment and FR_{local} is the local device measured value for each 100 m segment. The harmonization constants for the device are then a and b. The correlation coefficient of the regression and the standard error of estimate shall be reported.

Subsequent measurements made by the local friction device can be harmonized using the regression constants of the device:

$$IRFI = a + b FR_{local}$$
 (9)

Whenever the local friction device is modified, repaired or changes its calibration, new harmonization constants shall be determined.

It is foreseen that the propagated error for a two-stage harmonization, such as the standard error of estimates, will be up to twice the value of a one-stage harmonization.

As discussed in section 5.5, the prediction intervals are largely due to surface variability. Using a one-stage or two-stage transform may therefore not influence the error in calculated harmonized values heavily.

8.0 CONCLUSIONS AND RECOMMENDATIONS

The ASTM Standard E2100-00 defines and prescribes how to calculate IRFI for winter surfaces. The IRFI is a standard reporting index to provide information on friction characteristics of the movement area to aircraft operators.

The IRFI can be used by airport maintenance staff to monitor the winter frictional characteristics in support of surface maintenance actions.

The IRFI method typically reduces the present variations among different devices from 0.2 down to 0.05 friction units.

A reference device, which is required for calibration, must be a dedicated device for this purpose only, and the aviation community must agree on its provision, ownership and services. The device chosen for the exercises to demonstrate that IRFI is possible was an IMAG device called IRV. The IRV must be evaluated at some point for stability. If it is not stable with time, other references would need to be investigated. All harmonization constants will have to be reworked when a permanent IRFI reference has been designated.

There is proof that the participating devices in the JWRFMP are not representative of the other devices even when they are of the same generic type. This suggests that harmonization constants must be determined and applied to individual devices, rather than to generic groups of devices, as was done in the past and is the current paradigm in the aviation industry. To accomplish this, a master device can be calibrated to the IRFI reference device in order to serve as a secondary reference, and the manufacturer or owner of this secondary reference can then calibrate other devices to this master.

For any common scale of friction measure to work satisfactorily for the industry, annual harmonization meetings of devices must be arranged to determine the current harmonization constants, which will be valid only for a limited time: i.e. as long as the maintenance quality and product repeatability and durability will allow. The work in the JWRFMP so far has confirmed that friction devices do not report similar values for the

same surface and conditions unless they are harmonized on a regular basis, at least annually.

Further testing with a two-stage harmonization procedure is recommended to establish what differences in IRFI values a Master Reference harmonization method exhibits relative to harmonization with the IRFI Reference Vehicle.

Standardization of runway friction sampling techniques must be considered to avoid divergence in reported IRFI values due to differences in sampling techniques. There is a danger that two different IRFI regimes may evolve, one for continuous friction measuring equipment and one for spot measuring equipment. The goal of one common friction index worldwide would then not be fulfilled.

Preliminary work has shown that the IRFI can be correlated to various aircraft braking; this work needs to be completed.

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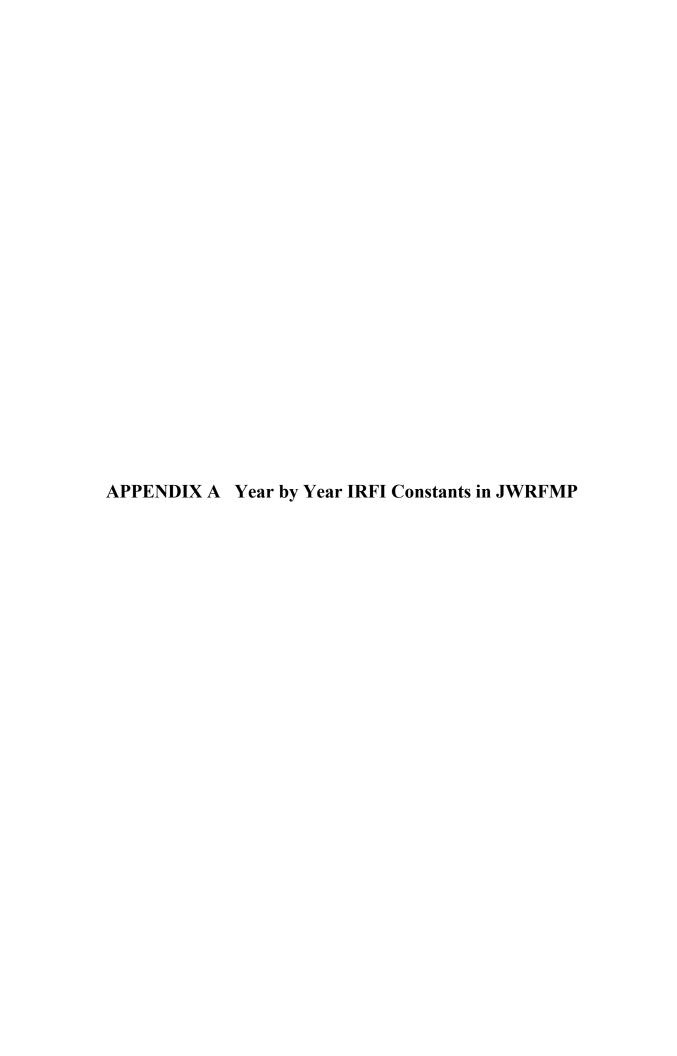


Table A.1 IRFI Constants for 1998

Device tire				StdError of	No of	
configuration	а	b	R^2	Estimate	datapoints	Comment
						Good plot, characteristic
ERD-NISSAN	0.03	0.90	0.45	0.076	176	outliers
ERD-23	0.08	0.83	0.72	0.042	83	
ASFT-ASFT-AERO-100	-0.05	0.91	0.78	0.072	536	
BV11-OSL-T520-100	0.03	0.75	0.85	0.061	528	
GRT-NCAA-						
SLUSHCUT	0.03	0.78	0.88	0.044	635	Good plot
GRT-NCAA-E1844-20	0.01	0.89	0.91	0.035	360	Good plot
SFT-TC79-E1551-100	0.05	0.71	0.95	0.034	683	Very good plot
						Two data clusters
						Good distribution in lower
BV11-196-T520-100	0.01	0.82	0.96	0.036	154	region
ITTV-NASA-AC26-136	0.08	1.01	0.96	0.037	141	
RFT-FAA-E1551-100	0.02	0.70	0.98	0.032	42	Two data clusters
SFT-212-E1551-100	0.02	0.74	0.99	0.023	42	Two data clusters

Table A.2 IRFI Constants for 1999

				StdError		
Device tire			_	of	No of	
configuration	а	b	R^2	Estimate	datapoints	Comment
BV11-OSL-T520-100	0.17	0.21	0.05	0.059	798	
SAR-MUN-AERO-100	0.14	0.32	0.11	0.062	678	
RFT-FAA-E1551-100	0.10	0.50	0.26	0.032	87	
ASFT-ASFT-AERO-						
100	0.10	0.49	0.37	0.052	607	
ERD-BLAZER	0.12	0.62	0.43	0.049	756	
ITTV-NASA-AC26-136	0.13	0.53	0.46	0.042	277	
ASFT-OSL-AERO-100	0.13	0.70	0.47	0.044	286	
SFT-TC79-E1551-100	0.09	0.88	0.63	0.040	1181	Even plot
GRT-NCAA-						
SLUSHCUT	0.02	0.93	0.65	0.031	432	
GRT-DND-E1844-20	0.02	0.94	0.67	0.036	490	Good, even plot
BV11-196-E1551-100	0.07	0.81	0.68	0.033	223	
GRT-NCAA-E1844-20	0.00	1.14	0.74	0.031	748	Good, even plot

Table A.3 IRFI Constants for 2000

Device tire				StdError of	No of data	
configuration	а	b	R^2	Estimate	points	Comment
GRT-IF-E1844-20	0.23	0.28	0.10	0.052	24	
ERD-BLAZER	0.16	0.56	0.43	0.096	286	
GRT-IF-SLUSHCUT	0.12	0.62	0.64	0.057	60	
SAR-MUN-AERO-100	0.19	0.38	0.65	0.058	55	
BV11-ZUR-T49-20	0.25	0.31	0.66	0.060	24	
SFT-TC85-E1551-100	0.17	0.59	0.68	0.121	49	
RFT-FAA-E1551-100	0.11	0.80	0.71	0.068	104	
ITTV-NASA-AC26-136	0.15	0.94	0.75	0.064	112	
ASFT-801-AERO-100	0.14	0.68	0.76	0.061	120	
BV11-196-E1551-100	0.14	0.61	0.78	0.090	91	
SFT-HAN-AERO-100	0.17	0.56	0.78	0.059	114	
SAR-813-AERO-100	0.15	0.57	0.79	0.061	102	
ASFT-USFT-AERO-100	0.04	0.99	0.79	0.084	97	
BV11-VIE-T520-100	0.16	0.56	0.81	0.056	108	
SFT-TC79-E1551-100	0.13	0.83	0.88	0.023	66	